

# MCE415

## Heat and Mass Transfer

Lecture 02: 18/09/2017

Dr. Ayokunle O. Balogun  
[balogun.ayokunle@lmu.edu.ng](mailto:balogun.ayokunle@lmu.edu.ng)

**Class: Monday (12 – 2 pm)**  
**Venue: B13**

# Etiquettes and MOP

- Attendance is a requirement.
- There may be class assessments, during or after lecture.
- Computational software will be employed in solving problems
- Conceptual understanding will be tested
- Lively discussions are integral part of the lectures.



# Lecture content

## Forced Convection

- Physical mechanism of convection
- Nusselt number
- Thermal boundary layer
- Prandtl number
- Parallel flow over flat plates

## Recommended textbook

- Fundamentals of Thermal-Fluid Sciences by Cengel Y.A., Turner R.H., & Cimbala J.M. 3<sup>rd</sup> edition

# Conceptual Understanding



What is the objective of this processes?

# Physical Mechanism of Convection Heat Transfer

- Convection heat transfer occurs between a surface and a fluid as a result of the presence of bulk motion in the fluid. In the absence of any bulk fluid motion then heat transfer is by conduction.
- Convection heat transfer is subdivided into *forced* or *natural* convection. Convection may also be viewed as *external* or *internal*

*Successive heat transfer enhancement by a blowing fan and replacement with water*

- Convection heat transfer depends primarily on the three factors highlighted below:
  1. Fluid properties
  2. Solid surface
  3. Fluid flow type
- A breakdown of the factors is shown on the succeeding slide

# Convection Heat Transfer

## Fluid properties

1. Dynamic viscosity,  $\mu$
2. Thermal conductivity,  $k$
3. Density,  $\rho$
4. Specific heat capacity,  $c_p$
5. Fluid velocity,  $V$

## Solid surface characteristics

1. Surface geometry
2. Surface roughness

## Fluid Flow Type

1. Streamlined (Laminar)
2. Turbulent

Fig 1: List of variables that affect convection heat transfer



# Physical Mechanism of Convection Heat Transfer

- Convection is the most complex of mechanism of heat transfer because of its dependence on many variables as shown in Fig 1.
- The complexity notwithstanding, the heat transfer rate is proportional to **temperature difference** and is expressed as **Newton's law of cooling** as:

$$\dot{q}_{conv} = h(T_s - T_{\infty}) \quad (W/m^2) \quad (1)$$

Or

$$\dot{Q}_{conv} = hA_s(T_s - T_{\infty}) \quad (W) \quad (2)$$

Where,

- $h$  = convection heat transfer coefficient,  $W/m^2 \cdot C$
- $A_s$  = heat transfer surface area,  $m^2$
- $T_s$  = temperature of the surface,  $^{\circ}C$
- $T_{\infty}$  = temperature of the fluid sufficiently far from the surface,  $^{\circ}C$

- **NOTE:**  $h$  depends on the several variables early mentioned and its determination is rather difficult.



# Physical Mechanism of Convection Heat Transfer

- The motion of fluid over solid surfaces assumes zero velocity at the point of contact with the surface due to viscous effects. This phenomenon is referred to as a **no-slip condition (NSC)**.
- The no-slip condition leads to the development of the **velocity profile**, i.e. variation in the velocity of adjacent layers of fluid flow.
- The region adjacent to the wall where this viscous effect is significant is called the **boundary layer** (Fig 2).
- The fluid property responsible for NSC and the development of the BL is **viscosity**.
- **Zero-velocity** at wall surface and **surface drag** are consequences of the NSC.

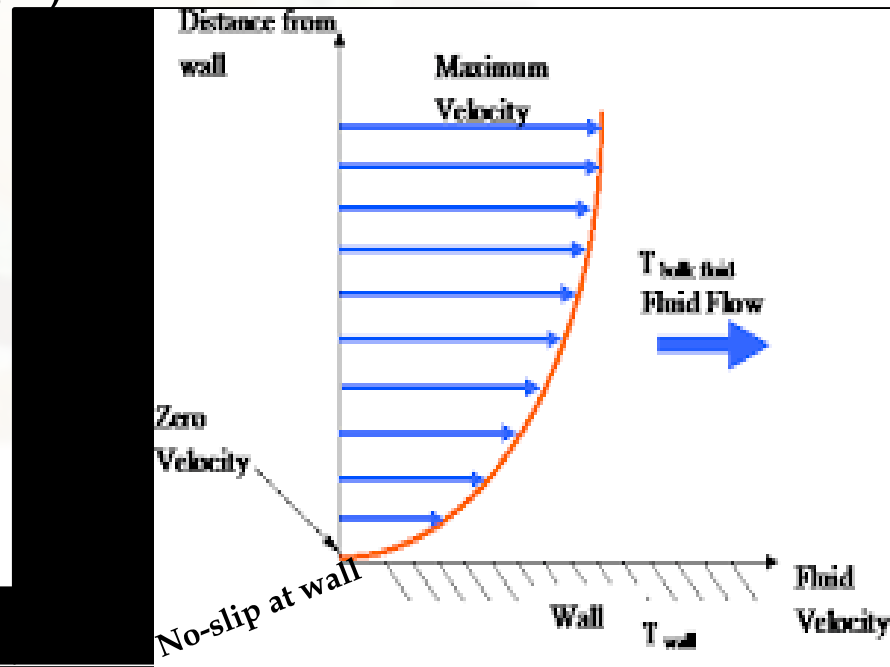


Fig 2: Velocity Profile of Boundary Layer



# Physical Mechanism of Convection Heat Transfer

- As a result of the NSC, heat transfer from the surface to the adjacent fluid layer is by pure conduction.

**WHY?**

This may be expressed as (Eq 3)

$$\dot{q}_{conv} = \dot{q}_{cond} = -k \left. \frac{\partial T}{\partial y} \right|_{y=0} \quad (W/m^2) \quad (3)$$

- $\therefore$  equating (1) and (3) for the heat flux yields (Eq 4)

$$h = \frac{-k(\partial T / \partial y)_{y=0}}{T_s - T_\infty} \quad (W/m^2 \cdot ^\circ C) \quad (4)$$

for the determination of the heat transfer coefficient when the temperature distribution within the fluid is known.

- The heat transfer coefficient,  $h$ , usually varies along the flow ( $x$ –) direction,  $\therefore$ , it is determined by averaging of the *local* heat transfer coefficient over the entire surface area  $A_s$  or length  $L$  as (Eq 5).

$$h = \frac{1}{A_s} \int_{A_s} h_{local} dA_s \quad \text{and} \quad h = \frac{1}{L} \int_0^L h_x dx \quad (5)$$

# Nusselt Number

- The **Nusselt number**,  $Nu$ , is a dimensionless quantity that measures the ratio of heat transfer by convection to conduction as expressed as (Eq 6)

$$Nu = \frac{hL_c}{k} \quad (6)$$

- The Nusselt number represents the enhancement of heat transfer by convection relative to conduction. Thus the higher  $Nu$  the more effective the convection
- An  $Nu = 1$  indicates pure conduction for the heat transfer across a the fluid layer.
- The enhancement of heat transfer by forced convection in daily life are varied and they include:

# Thermal Boundary Layer

- Recall the development of the velocity boundary layer, which is defined as the region in which the fluid velocity varies from zero to  $0.99V$ .
- Similarly, a thermal boundary layer develops as shown in Fig 2.
- The flow region over the surface in which the temperature variation in the direction normal to the surface is significant is the **thermal boundary layer (TBL)**.
- The *thickness*,  $\delta_t$ , of the TBL at any location along the surface is *the distance from the surface at which the temperature difference  $T - T_s$  equals  $0.99(T_\infty - T_s)$* .

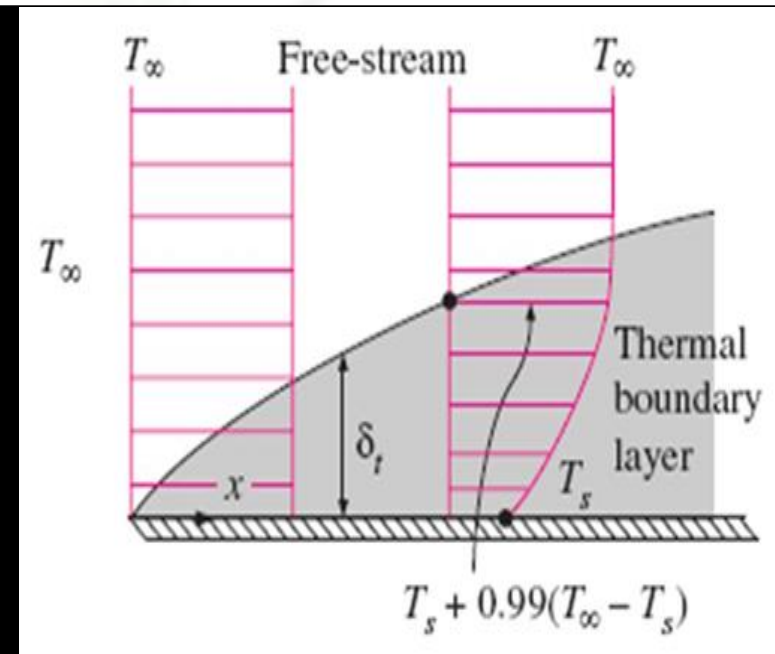


Fig 3: Thermal boundary layer on a flat plate (the fluid is hotter than the plate surface)

# Salient points on TBL & VBL

What would  $T$  be for a special case of  $T_s = 0$ ? How does it relate to the velocity boundary layer,  $u$ ?

- The *thickness*,  $\delta_t$ , of the TBL increases in the flow direction, since the effects of heat transfer are felt at greater distances from the surface further down stream
- The convection heat transfer rate anywhere along the surface is directly related to the temperature gradient at that location.
- Therefore the shape of the temperature profile in the TBL dictates the convection at transfer between the solid and the fluid flowing over
- In a flow over a heated (cooled) surface, both the VBL and TBL develop simultaneously.
- Note that fluid velocity influences the development of the temperature profile, therefore the development of the VBL relative to the TBL will have a strong effect on the convection heat transfer.



# Prandtl Number

- The relative thickness of the velocity and the thermal boundary layer is defined by the dimensionless parameter **Prandtl number**,  $Pr$ , as (Eq 7)

$$Pr = \frac{\text{Molecular diffusivity of momentum}}{\text{Molecular diffusivity of heat}} = \frac{\nu}{\alpha} = \frac{\mu c_p}{k} \quad (7)$$

- The  $Pr$  for gases is about 1. This indicates that both momentum and heat dissipate through the fluid at about the same rate.
- Heat diffuses quickly in liquid metals ( $Pr \ll 1$ ) and very slowly in oils ( $Pr \gg 1$ ), relative to momentum.

**Consequently *the TBL is much thicker for liquid metals and much thinner for oils relative to the VBL.***

## Typical ranges of Prandtl numbers for common fluids

Fluid	Pr
Liquid metals	0.004 - 0.030
Gases	0.7 - 1.0
Water	1.7 - 13.7
Light organic fluids	5 - 50
Oils	50 - 100,000
Glycerin	2,000 - 100,000

# Parallel Flow Over Flat Plates

- Transition from laminar to turbulent regions in the velocity boundary layer during a flow over a flat plate is as shown in Fig 4.
- This transition depends on *surface geometry*, *surface roughness*, *upstream velocity*, *surface temperature*, and *type of fluid* among other things. The transition is best characterized by Reynolds number at a distance  $x$  from the leading edge.

- The Reynolds number at a distance  $x$  from the leading edge of a flat plate is expressed as

$$Re_x = \frac{\rho V x}{\mu} = \frac{V x}{\nu} \quad (8)$$

Note,  $Re$  varies along  $x$  direction reaching a  $Re_L = VL/\nu$  at the end of the plate

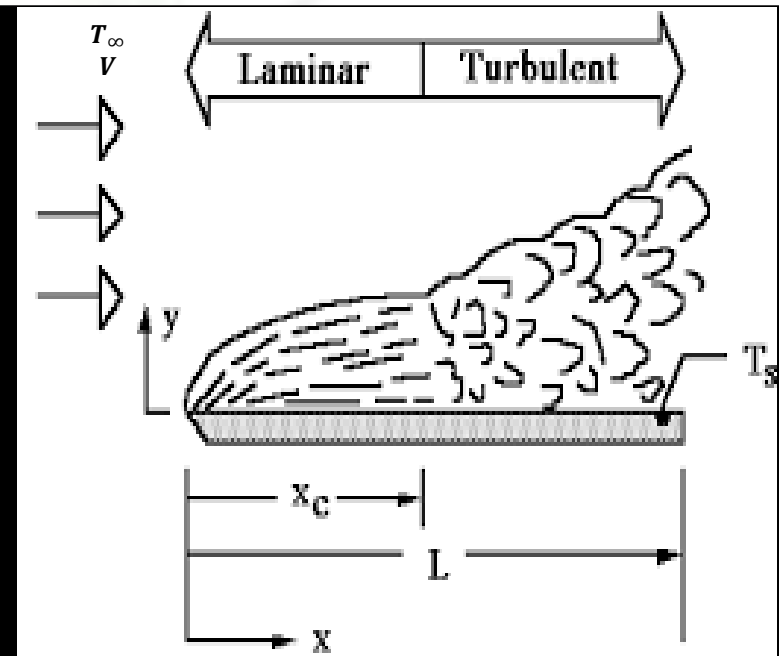


Fig 4: Laminar and turbulent regions of the boundary layer during flow over a flat plate



# Parallel Flow Over Flat Plates

- To establish the critical point of transition between laminar to turbulent regions in engineering analysis, the critical Reynolds number is usually set at Eq 9

$$Re_{cr} = \frac{\rho V x_{cr}}{\mu} = 5 \times 10^5 \quad (9)$$

- Though the actual value of engineering critical Reynolds number for a flat plate may vary from  $10^5$  to  $3 \times 10^6$ , depending on the *surface roughness, turbulence level, and the variation of pressure* along the surface.

# The local Nusselt number

- The *local* Nusselt number at a location  $x$  for laminar flow over a flat plate is expressed as Eq 10

$$\text{Laminar: } Nu_x = \frac{h_x x}{k} = 0.332 Re_x^{0.5} Pr^{1/3} \quad Pr > 0.6 \quad (10)$$

- and the corresponding turbulent flow is Eq 11

$$\text{Turbulent: } Nu_x = \frac{h_x x}{k} = 0.0296 Re_x^{0.8} Pr^{1/3} \quad 0.6 \leq Pr \leq 60 \quad (11)$$

$$5 \times 10^5 \leq Re_x \leq 10^7$$

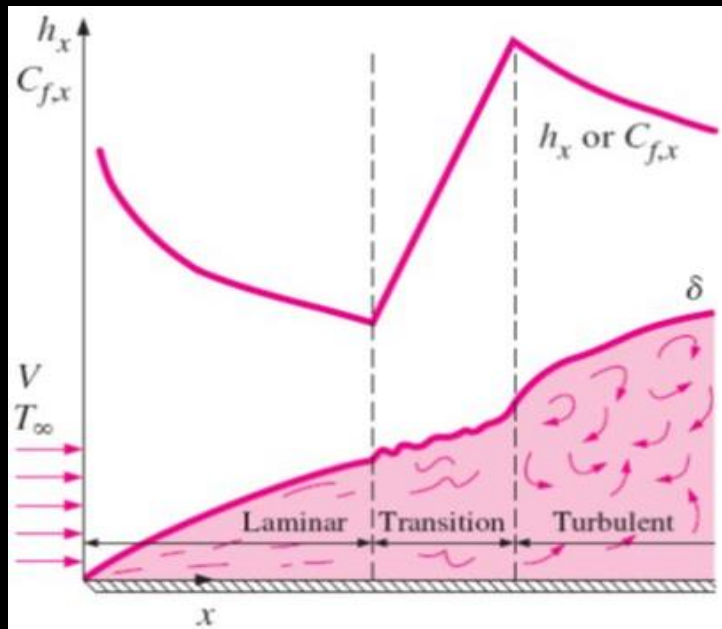


Fig 5: The variation of the local friction & heat transfer coefficients for flow over a flat plate

- Note the  $h_x \propto Re_x^{0.5}$  and thus  $\propto x^{-0.5}$  for laminar flow.
- $\therefore h_x \rightarrow \infty$  at leading edge ( $x = 0$ ) &  $\downarrow_{ses}$  by a factor of  $x^{-0.5}$  in the flow direction.
- In Fig 5, the local  $C_{f,x}$  &  $h_x$  reaches maximum when flow becomes fully turbulent and  $\downarrow_{ses}$  by a factor of  $x^{-0.2}$

# The average Nusselt number

- The *average* Nusselt number over the entire is obtained putting Eqs 10 & 11 into Eq. 5 to yield Eq 12

$$\text{Laminar: } Nu = \frac{hL}{k} = 0.664Re_L^{0.5}Pr^{1/3} \quad Re_L < 5 \times 10^5 \quad (12)$$

- and the corresponding turbulent flow is Eq 13

$$\text{Turbulent: } Nu = \frac{hL}{k} = 0.037Re_x^{0.8}Pr^{1/3} \quad 0.6 \leq Pr \leq 60 \quad (13)$$

$$5 \times 10^5 \leq Re_L \leq 10^7$$

- Eq 12 represents the average heat transfer coefficient for the entire plate when the flow is *laminar* over the *entire* plate.
- Eq 13 is for the flow that is *turbulent* over the entire plate, or when the laminar flow region is too small relative to the turbulent flow.
- Where both laminar and turbulent flows contribute significantly, Eq. 14 is the average Nusselt number applicable

$$Nu = \frac{hL}{k} = (0.037Re_L^{0.8} - 871)Pr^{1/3} \quad 0.6 \leq Pr \leq 60 \quad (14)$$

$$5 \times 10^5 \leq Re_L \leq 10^7$$



# The Nusselt number

- A single correlation that applies to all fluids, including metal liquids have been proposed by Churchill and Ozoe (1973) as Eq 15

$$Nu_x = \frac{h_x x}{k} = \frac{0.3387 Pr^{1/3} Re_x^{0.5}}{[1 + (0.0468/Pr)^{2/3}]^{1/4}} \quad (15)$$

- These relations are for *isothermal* surfaces but could also be used for non-isothermal surfaces by assuming constant surface temperature at some average value.
- The assumptions for these relation includes
  1. Smooth surface, and
  2. Free stream region is turbulent free
- The effect of variable properties can be accounted for by evaluating properties at the film temperature

# Flat Plate with Unheated Starting Length

- In cases where the starting section is unheated as shown in Fig 6, which means no heat transfer at  $0 < x < \xi$
- This implies that the VBL starts to develop at the leading edge ( $x = 0$ ), while the onset of the TBL is where heating starts ( $x = \xi$ ).
- For flat plate with heated section maintained at a constant temperature ( $T = T_s$  constant for  $x > \xi$ )
- Using integral solution methods, the local Nu respectively for laminar and turbulent is Eq 16 & 17 for  $x > \xi$

$$Nu_x = \frac{Nu_{x(\text{for } \xi=0)}}{[1-(\xi/x)^{3/4}]^{1/3}} = \frac{0.332Re_x^{0.5}Pr^{1/3}}{[1-(\xi/x)^{3/4}]^{1/3}} \quad (16)$$

$$Nu_x = \frac{Nu_{x(\text{for } \xi=0)}}{[1-(\xi/x)^{9/10}]^{1/9}} = \frac{0.0296Re_x^{0.8}Pr^{1/3}}{[1-(\xi/x)^{9/10}]^{1/9}} \quad (17)$$

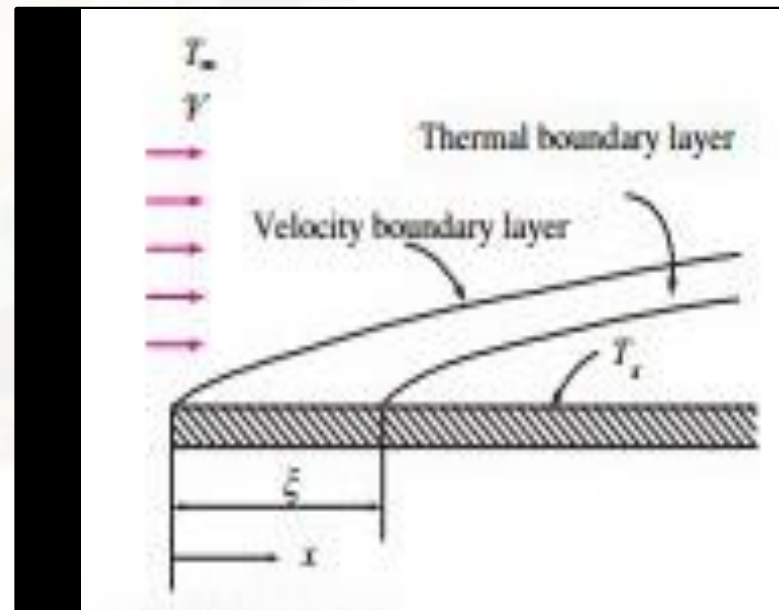


Fig 6: Flow over a flat plate with an unheated starting length

# Flat Plate with Unheated Starting Length

- The determination of the average Nusselt number for the heated section of the plate is obtained by integrating Eqs 16 & 17. This can not be done analytically, therefore, the numerical integration yields Eqs 18 & 19 respectively for laminar and turbulent flow

$$\text{Laminar:} \quad h = \frac{2[1-(\xi/x)^{3/4}]}{1-\xi/L} h_{x=L} \quad (18)$$

$$\text{Turbulent:} \quad h = \frac{5[1-(\xi/x)^{9/10}]}{4(1-\xi/L)} h_{x=L} \quad (19)$$

- Eq 18 is the average heat transfer coefficient for the entire heated section of the plate when flow is laminar over the entire plate. Note that for  $\xi=0$  it reduces to  $h_L = 2h_{x=L}$ , as expected.
- Eq 19 is the average convection coefficient for the case of turbulent flow over the entire plate or when the laminar region is small relative to the turbulent region.





# Uniform Heat Flux

- When a flat plate is subjected to uniform heat flux instead of uniform temperature, the local Nusselt number is given as

$$\text{Laminar:} \quad Nu_x = 0.453 Re_x^{0.5} Pr^{1/3} \quad (20)$$

$$\text{Turbulent:} \quad Nu_x = 0.0308 Re_x^{0.8} Pr^{1/3} \quad (21)$$

- These relations give values that are 36% higher for laminar flow and 4% higher for turbulent flow relative to the isothermal case
- When the plate involves an unheated starting length, the relations in Eqs 16 & 17 can still be used provided that Eqs 20 & 21 are used for  $Nu_x$  respectively.
- When heat flux  $\dot{q}_s$  is prescribed, the rate of heat transfer to or from the plate and the surface temperature at a distance  $x$  are determined as

$$\dot{Q} = \dot{q}_s A_s \quad (22)$$

$$\dot{q}_s = h_x [T_s(x) - T_\infty] \Rightarrow T_s(x) = T_\infty + \dot{q}_s / h_x \quad (23)$$



# Example

1. Engine oil at  $60\text{ }^{\circ}\text{C}$  flows over the upper surface of a  $5\text{ m}$  long flat plate whose temperature is  $20\text{ }^{\circ}\text{C}$  with a velocity of  $2\text{ m/s}$ . determine the rate of heat transfer per unit width of the entire plate.
2. The local atmospheric pressure in Denver, Colorado (elevation  $1610\text{ m}$ ) is  $83.4\text{ kPa}$ . Air at this pressure and  $20\text{ }^{\circ}\text{C}$  flows with a velocity of  $8\text{ m/s}$  over a  $1.5\text{ m} \times 6\text{ m}$  flat plate whose temperature is  $140\text{ }^{\circ}\text{C}$ . Determine the rate of heat transfer from the plate if the air flows parallel to the (a)  $6\text{ m}$  long side (b) the  $1.5\text{ m}$  side.
3. Repeat example 2a, using MATLAB, plot the rate of heat transfer against velocity of air from  $1 - 17\text{ m/s}$ . Discuss the result.



# Assignment

Fundamentals of thermal-fluid sciences by Cengel Y.A., Turner R.H. & Cimbala J.M. 3<sup>rd</sup> Edition

[From the above textbook, PP 892-893, answer 1-2](#)

1. Question 19-1C to 19-6C
2. Question 19-12C, 19-16 to 19-17



Forced convection